Airborne Geophysical Survey

Heemskirk NW, Tasmania

March 2007 Survey Operations and Logistics Report For STELLAR RESOURCES LIMITED

Survey Flown by:



GPX Airborne

Job 2273

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1 GENERAL SURVEY INFORMATION

1.1 INTRODUCTION

In February 2007, GPX Airborne commenced a fixed wing airborne magnetic and radiometric survey for Stellar Resources Limited in the Heemskirk area in Tasmania. The survey was flown using a Cessna 210 owned and operated by Ozshore. This report summarizes the procedures, details and equipment used by GPX Airborne in the acquisition, verification and processing of the airborne geophysical data.

Client:	Stellar Resources Limited
GPX Job Number:	2273
Survey Area:	Heemskirk NW, Tasmania.
Data Processing Base:	Strahan Comfort Inn, Tasmania.
Mobilisation:	23 rd February 2007
Production:	26 th February to 5 th March 2007
De-Mobilisation:	5 th March 2007
	5 th March 2007 2414 Kms

1.2 SURVEY BRIEF

Aircraft equipment installation, test and radiometric calibration flights were carried out in the month of August 2006 in Perth, Western Australia. System stability and continuity had been monitored throughout all surveys. The crew arrived on the 23rd February and commenced test flying on the 26th February. The crew stayed at the Strahan Comfort Inn, where all survey operations were run from and where the post flight data verification was performed. The base magnetometer was also set up near the airport. The final flight of the survey was on the 5th March 2007.

1.3 SURVEY PERSONNEL

The following personnel were involved on this project:

Operations and Safety Manager:	Bob Blizzard
Project Leader	Jeff Ibbotson
Technical Support:	Mike Barrett
Operator:	Don Copley
Pilots:	Guy Nash Vincent Wong
Data Processing:	Cathy Carr

1.4 SURVEY EQUIPMENT

Survey Platform	
Data Acquisition System	
Magnetometer Processor	
Magnetometer	
Spectrometer	
Fluxgate Magnetometer	
GPS Receiver	
DGPS Receiver	
Radar Altimeter	
Magnetic Base Stations	
In-field Computer	
In-field Software	

Cessna 210 (VH-MNN). Pico Envirotec AGIS PC104 Console. Pico Envirotec MMS4 Magnetometer Processor Geometrics G-822A Cesium Vapour Exploranium GR820 (32 Litre Crystal) Billingsley TFM100-G2 Novatel OEM3 Propak II CSI DGPSMax Collins ALT-50A Envirotec G-mag Cesium vapour. Toshiba Notebook Pico Envirotec PEIView, ChrisDBF

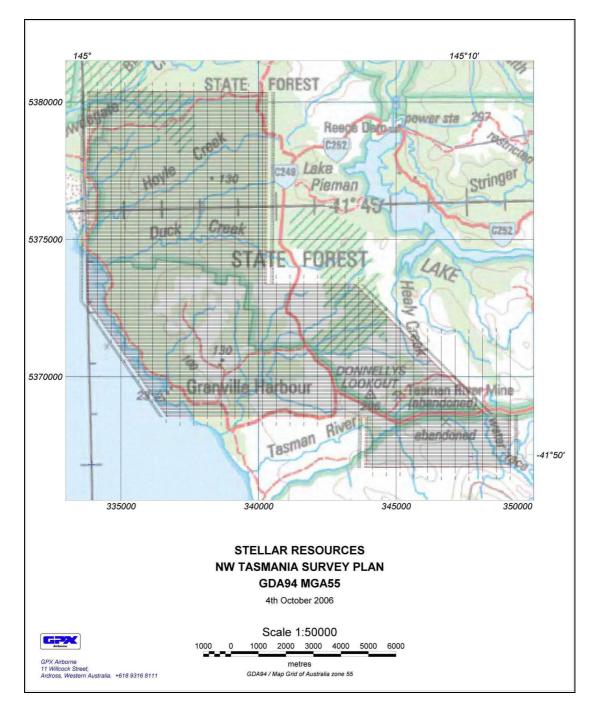


VH-MNN

1.5 SURVEY AREA

The following coordinates are in GDA94 / Map Grid of Australia zone 55 and defines the survey area.

Easting	Northing
333738	5372731
333782	5380476
340352	5380431
340308	5373450
343510	5373452
348251	5368710
349178	5368708
349176	5366662
343834	5366617
343878	5368461
336673	5368463



Survey Plan on Australian Topography

1.6 SURVEY PARAMETERS

Line spacing: Line direction: Tie line spacing: Tie line direction: Minimum line length: Sensor height: Magnetometer sample rate: Spectrometer sample rate: Altimeter sample rate: Base magnetometer sample rate: 50 metres 90° and 270° 500 metres 000° and 180° 5000 metres 60 metres 10 Hz 1 Hz recording 256 channels 1 Hz 1 Hz

2 SURVEY EQUIPMENT SPECIFICATIONS

2.1 DATA ACQUISITION CONSOLE

The Data Acquisition console is a Pico Envirotec AGIS PC104. This is a versatile multi-function system that is capable of operation in many different configurations, depending on platform type, navigation and system requirements. The AGIS PC104 provides the following functions:

- Navigation / flight control
- Data recording
- Display of real-time collected data and status monitoring
- Data retrieval access



Real time monitor and navigation console.

2.1.1 Navigation / Flight Control

The AGIS PC104 is used to guide the aircraft on a pre-defined flight plan that can be generated in UTM or Latitude/Longitude coordinates. The pre-defined flight plan can be designed to file prior to the start of the project, entered or altered in the AGIS system or delineated 'on-the-fly' e.g. while in the air flying the boundary and entering corner coordinates. Co-ordinates can only be entered in the WGS84 datum system, this has been implemented to avoid confusion and eliminate possible conversion errors. Normal survey altitude and ground speed, with pre-set tolerances are also entered.

The pilot display consisted of a 2-line strip display or more comprehensive Pilot Guidance Unit (PGU). The strip display is driven directly from the AGIS

PC104 console; whereas the PGU is a self-contained computer system that is capable of more demanding navigation functions such as "drape" flying using a pre-programmed altitude grid.

The desired flight line is selected from the operator interface, which will either be a keyboard or touch-screen.

2.1.2 Data Recording

The AGIS PC104 relates all acquired data to the instant position from the GPS receiver and records the collected data to three separate data files. The data is recorded in compressed binary format, to a commercial solid-state hard disk.

The flight path file is recorded from AGIS program start-up to shutdown and cannot be turned off by the operator. It contains position, timing, altitude and basic data.

The data file is recorded whenever the acquisition system is "On-line". It contains all navigation data plus "enabled" data.

The raw data file, when enabled and supported by the GPS receiver in use, contains raw GPS data necessary for post-flight position correction. It is recorded from AGIS program start-up to shutdown.

2.1.3 Display of real-time collected Data and status monitoring

The AGIS displays flight path and geophysical data as it is acquired aiding the data quality control and real time navigation guidance. The user is presented with graphical representations of the survey area, flight lines, navigation status, and sensor data. The spectra data was also displayed.

Several other status indications are also provided which will either change state indicating a major system malfunction, such as a magnetometer or spectrometer failure, or will change state during normal operation, indicating data being written to a file etc

2.1.4 Data Retrieval

The AGIS PC104 provides facility to transfer the recorded data from the internal solid-state disk to compact flash media immediately following the completion of the survey flight. Recorded data is not deleted from the main disk until this "retrieved" data has been verified "error free".

2.2 MAGNETOMETER PROCESSOR

The Magnetometer Processor is a Pico Envirotec MMS4 Magnetometer Processor. This is an advanced frequency-measuring device that can support several continuous signal magnetometers (Cs, He, K). It is a hardware-software designed system, exhibiting simplicity, easy interfacing and

substantial versatility. Magnetometer readings are synchronized with the PPS (Pulse Per Second) signal derived from the GPS for accurate timing.

The MMS4 contains 8 channels of analog differential inputs. The first 4 analog channels are sampled synchronously with MMS4 magnetometer at up to 50 samples per second. The remaining 4 analog channels are sampled at 10 samples per second. Analog data is integrated into the magnetometer data stream.

Specifications:

Input:	Coaxial - Larmour signal over DC Power Supply
Resolution:	0.0002 nT (Gamma) = 0.2 picoTesla
Sampling rates:	10, 20, 50 samples per second
Dynamic range:	15000 to 100000nT
Synchronization:	GPS – PPS (Pulse Per Second)
Data Storage:	Removable Compact Flash Memory

2.3 MAGNETOMETER SENSOR

The Magnetometer Sensor is a Geometrics G-822A, which employs an optically pumped cesium-vapour atomic magnetic resonance system that function as the frequency control element in an oscillator circuit.

Specifications:	
Model:	Geometrics G-822A
Operating Range:	20,000 – 100,000 nT
Sensitivity:	Typically 0.002 nT P-P at a 10Hz sample
	rate
Heading Error:	< 0.15 nT over entire 360°
Output:	Larmour frequency, 3.498572 Hz/nT

2.4 FLUXGATE MAGNETOMETER

The Fluxgate Magnetometer is a Billingsley Ultra Miniature TFM 100G2. This unit is a low noise, high sensitivity unit, packaged into a compact housing. An analog DC output voltage is produced for each of the measured X, Y and Z orthogonal components of the current magnetic field.

Specifications:

Model:	Billingsley TFM 100G2
Axial Alignment:	Orthogonality better than ±1°
Sensitivity:	100uV / nT
Noise:	20pT RMS / Hz @ 1Hz
Output:	$\pm 100 uT = \pm 10 V$

2.5 SPECTROMETER

The Spectrometer is an Exploranium GR820 system. The unit comprises of 2 detector crystal packs, which give a total volume for detection of 32 litres. The spectrometer employs automatic gain stabilisation control to eliminate the need to heat the detectors. Signal processing automatically perform digital gain control to the individual crystal spectra, ensuring the summed spectrum is stable.

Model:	Exploranium GR820
Sensitivity:	0-3.0 MeV
Maximum count rate:	100,000 counts/sec
Detector volume:	16.7 Litres (each)
Detector weight:	83.9 kgs (each)

2.6 TEMPERATURE AND HUMIDITY SENSORS

The Temperature and Humidity transmitter is a Vaisala HMP233. The unit provides both a digital RS232 output and Analogue voltage or current output directly proportional to the measured Temperature and Humidity. The unit is a commercial grade device housed in a rugged aluminium enclosure.

Specifications:	
Model:	HMP233
Humidity Range:	0 – 100% RH
Humidity Accuracy:	±1 %RH (090 %RH)
	±2 %RH (90100 %RH)
Temperature Range:	-40 to +80°C
Temperature Accuracy:	± 0.1°C
Analog Output Accuracy:	± 0.05 % full scale

2.7 BAROMETRIC PRESSURE SENSOR

The Barometric Pressure transmitter is a Vaisala PTB220. The unit provides both a digital RS232 output and Analogue voltage or current output directly proportional to the measured Barometric Pressure. The unit is a Class "A" commercial grade device housed in a rugged aluminium enclosure.

Specifications:

Model:	PTB220
Range:	500 – 1100 hPa
Resolution:	0.01 hPa
Accuracy at +20°C:	\pm 0.1 hPa

2.8 RADAR ALTIMETER

The Radar Altimeter is a Rockwell Collins ALT-50 two-antenna unit operating at a centre frequency of 4300MHz. The voltage output to the data system is directly proportional to the aircraft flying height with an output characteristic of 20mV/ft up to 500ft, then 10.4V + 3mV/ft above 500ft.

Specifications:	
Model:	Collins ALT-50A Radio Altimeter System
Accuracy:	\pm 3ft - 0 to 150ft range
	\pm 2% of indicated altitude – 150 to 500ft range
	\pm 3.5% of indicated altitude – 500 to 200ft range
Measurement Rate:	Same rate as magnetometer, 10Hz minimum.

2.9 GPS RECEIVER

The GPS receiver is a Novatel OEM3 GPS Receiver housed in a Propak II enclosure. The antenna is a Rojone wideband unit, which has it's output split to provide signal for both the GPS and DGPS receivers.

Specifications:	
Receiver:	CSI DGPS MAX
Position update rate:	5Hz
Raw data update rate:	1Hz
Input frequency:	L1
Antenna:	Fugro Wideband – Stinger Mounted

2.10 DGPS RECEIVER

The DGPS receiver is a CSI DGPS MAX, which is a 12-channel combined GPS/DGPS unit. The DGPS MAX is able to use differential corrections received through an internal WAAS demodulator, VLF beacon receiver, or the OmniSTAR DGPS Service.

Specifications:

CSI DGPS MAX
5Hz
L1
Fugro Wideband – Stinger Mounted
Typically every 6 seconds
OmniSTAR VBS

2.11 BASE MAGNETOMETER

Pico Envirotec G-mag base magnetometer.

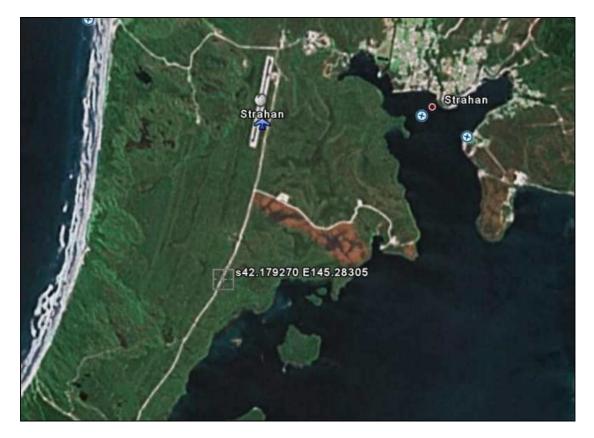
Specifications:

Type: Resolution: Absolute Accuracy: Dynamic Range: Sampling Rate: Data Storage: G-Mag Cesium Vapour 0.001nT 0.1nT 20000 to 100000 nT 10Hz Flash disk.

2.11.1 Base Station Location

The base station was located at

Longitude: 145.28305 E Latitude: 42.17927 S



Location map

3 EQUIPMENT CALIBRATIONS AND DATA ACQUISITION CHECKS

3.1 DYNAMIC MAGNETOMETER COMPENSATION

Aircraft compensation tests were flown at high altitude on the 4 survey line headings and also at +/-15° to the line headings (to accommodate for cross wind flying conditions). The data for each heading consists of a series of aircraft manoeuvres with large angular excursions: specifically pitches, rolls and yaws. This is done to artificially create the worst possible attitudes and rates of attitudinal change likely to be encountered while on line and compensate for any magnetic noise created by the aircraft's motion within the earth's magnetic field. This data is processed to obtain the REAL TIME COMPENSATION terms of which the aircraft used the standard 17-term model. These terms include permanent, induced and eddy values. These coefficients may be applied in real time or during post processing. Note that this form of compensation will only remove those noise effects modelled in the manoeuvres test flight. External noise sources and random motions of the stinger with respect to the aircraft airframe generally establish the noise floor for this type of installation. The surveyor's goal is to achieve a 4th difference noise level on the order of 0.01nT RMS during normal surveying conditions. In general, this noise level was routinely achieved or bettered as a matter of course.

3.2 HEADING ERROR CHECK

Historically, heading error checks have been an essential part of the aeromagnetic data acquisition procedure but their importance now has diminished. GPX Airborne now corrects for these effects using the dynamic aircraft magnetic compensation system and specially developed software. In the past, repeatable heading errors of less than one nanotesla (1.0nT) were considered good. Dynamic compensation typically yields heading errors in the order of 0.1 to 0.3 nT, which are effectively eliminated by modern data levelling techniques.

3.3 SYSTEM PARALLAX TESTS

One of the processing parameters required to process digital data was the parallax or offset time, between the time the digital reading was taken by the instrument and the time the position fix for the fiducial of the reading was obtained. Each instrument - magnetometer, altimeter - may have a different parallax, so the parallax must be computed for each instrument.

The parallax correction derived is the correction to be applied to each survey line. A positive parallax indicates the instrument reading is ahead of the position of the fiducial. Each integer fiducial represents one second so the parallax can be expressed in either fiducial or seconds.

The correct fiducial is computed by:

Parallax corrected fid = Fid for recorded reading – Instrument parallax

Results of parallax test.

Channel	Parallax applied
Position	0.8
Magnetic	1.8
Radiometric data	0.0
Radar Altimeter	1.8
Barometric Altimeter	1.8
GPS derived DEM	0.0
Barometric derived DEM	0.0
Temperature	1.8
Pressure	1.8
Humidity	1.8
GPS Altimeter	0.8
Fluxgates	0.0

3.4 ALTIMETER CALIBRATIONS

The height of the aircraft above ground is recorded by a radar altimeter as a voltage every 0.1 second. The voltage data is converted to height via a lookup table determined by calibration with the GPS altitude.

3.5 RADIOMETRIC PRE SURVEY CALIBRATIONS

The results of radiometric calibrations conducted by flying over the Yarra Yarra salt lake, and the Carnamah test range in Western Australia. The Pad tests were completed in August 2006 at the Jandakot Airport.

Data acquisition was by GPX Airborne, using a Cessna 210, VH-MNN. The total crystal volume was 33 litres.

The Calibration methods are as generally described by Grasty and Minty (1995).

VH-MNN	Date	Window	Value
Aircraft Background	9-August 2006	TC	81.67
_		K	23.36
		U	0.59
		Th	0.67
Cosmic Background	9-August 2006	TC	0.811201
		K	0.044663
		U	0.038646
		Th	0.043791
Stripping	1-August 2006	Alpha	0.2503
		Beta	0.3958
		Gamma	0.7550
		а	0.0571
		b	-0.0089
		С	-0.0041
Height Attenuation	4-August 2006	TC	0.006928
		K	0.009043
		U	0.007186
		Th	0.006892
Air/Ground @60m	4-August 2006	Dose	31.243915
		K	113.982381
		U	10.5493357
		Th	6.47931431
Air/Ground @80m	4-August 2006	Dose	27.2012973
		K	95.1242451
		U	9.13691585
		Th	5.64491245

A summary of the results is shown in Table 1 below.

3.6 DAILY RADIOMETRIC CHECKS

A system stability test was performed at the start and end of each day. This was done using a thorium source placed a least 40cm from the centre of each detector. The average deadtime and background corrected thorium window was calculated and checked to be within 3 percent from the average of all other calibrations.

3.7 DAILY TIME SYNCHRONIZATION

Before each days survey the magnetic base station is automatically synchronized with the GPS receiver time in the aircraft. Prior to the commencement of survey, the temporal drift of this base station was determined. The unit is automatically updated by the GPS so there is no time drift in the system.

3.8 SURVEY LINE NUMBERING SYSTEM

The first digit in any line number represents the area number, i.e. 100050 is area no. 1.

The next four numbers are the line number it self, i.e. 101030 is line number 103.

All Tie lines begin with the digit 7, i.e. 170020.

The sixth digits of any line number represent the attempt number, i.e. 100010 is the first attempt.

4 DATA VERIFICATION AND FINAL PROCESSING

4.1 IN FIELD DATA PROCESSING

All data verification and preliminary processing and map production was conducted at the field office using a Toshiba Notebook computer. ChrisDBF was the primary field quality control software.

At the conclusion of each days survey all magnetic, radiometric, altimeter, flight path and diurnal data was transferred via compact flash memory onto the office computer for preliminary data verification.

4.1.1 Altimeter Data

Radar Altimeter Data

The radar altimeter is verified to check that a reasonably constant height above the terrain specified in section 1.7 was flown; readings during the course of the survey did not exceed the specified tolerances. The radar altimeter data is used in the production of digital terrain maps.

GPS Height Data

The aircraft's height above mean sea level each second was determined by data from the post-processed GPS. The GPS height of the aircraft is verified to check for data masking and for equipment reliability. The GPS height data is used in the production of digital terrain maps.

Digital Terrain Data

After verification the radar altimeter height was subtracted from the GPS height to give the elevation of the terrain above mean sea level.

Gridding and Inspection

The digital terrain data was gridded and grid image enhancements were computed and displayed on screen. These were viewed also with the aid of crossline sun angles and inspected for inconsistencies and errors and appropriate corrections were made if required.

4.1.2 Flight Path Data

The flight path is plotted daily to ensure it was within survey specifications. Any data not within specification was re-flown. The aircraft GPS recorded the data in the WGS84 datum.

4.1.3 Magnetic Data

The raw un-edited magnetic data was checked to identify noise and spikes. Single reading spikes were manually edited and if the noise exceeded the contract specifications, the line was re-flown.

Magnetic Diurnal Data

Diurnal data recorded every 0.1 second from the primary base station was down loaded from the magnetometer's memory onto the field processing computer via compact flash. The diurnal data was then checked and corrected for spikes. Single reading spikes were manually edited and multiple erroneous readings flagged as invalid. If invalid diurnal data occurred whilst survey data was being acquired the affected section was re-flown. The diurnal data was also checked to see that the change in diurnal readings during the course of the survey did not exceed the specified tolerances. When this occurred the affected survey lines were re-flown. The diurnal data was merged with the aircraft data and used in the verification of the magnetic data.

Diurnal Correction

The synchronized digital diurnal data collected by the base station was first subtracted from the corresponding airborne magnetic readings to calculate a difference. The resultant difference was then subtracted from the base value to produce diurnally corrected magnetic data.

Parallax Correction

The aircraft system parallax is also checked prior to project commencement. A parallax error correction of 0.0 second was used for in field verification.

Gridding and Inspection

The magnetic data was gridded and grid image enhancements were computed and displayed on screen. These were also viewed with the aid of crossline sun angles and inspected for inconsistencies and errors and appropriate corrections were made if required.

4.1.4 Radiometric Data

Spectra Verification

The 256-channel radiometric data is viewed to confirm that the spectra peaks are correctly calibrated. The following peak locations are checked daily.

- Potassium 1460 keV
- Uranium 1760 keV
- Thorium 2614 keV

Parallax Correction

The aircraft system parallax is also checked prior to project commencement. A parallax error correction of 0.0 second was used for in field verification.

Gridding and Inspection

The radiometric data was gridded and grid image enhancements were computed and displayed on screen. These were also viewed with the aid of crossline sun angles and inspected for inconsistencies and errors and appropriate corrections were made if required.

4.1.5 Digital Archives

All raw aircraft, and diurnal base data were backed up on CD-ROM disk at the end of each day's survey. A further backup of all raw and edited data remained on the field-processing computer for the entire duration of the project. A copy of each days flying was transferred to the company's ftp site for further verification.

4.2 FINAL PROCESSING

All final data processing of the data was performed in the offices of GPX Airborne. Raw field data was transferred to the offices and processed to produce the final data. No field-processed data was used in the making of the final data. The final processing of the data follows the same quality control checks that are made in the field, however the final data has additional processes performed.

4.2.1 Altimeter Data

Radar Altimeter Data

The radar altimeter is verified to check that a reasonably constant height above the terrain specified in section 1.7 was flown; readings during the course of the survey did not exceed the specified tolerances. The radar altimeter data is used in the production of digital terrain maps.

GPS Height Data

The aircraft's height above mean sea level each second was determined by data from the post-processed GPS. The GPS height of the aircraft is verified to check for data masking and for equipment reliability. The GPS height data is used in the production of digital terrain maps.

Parallax Correction

A parallax error correction as described in section 3.3 was applied to the coordinate data.

Tie Line Levelling

A crossover program was used to compute the height difference between each tie line and the traverse line intersection. These differences were then applied to level the traverse lines to the tie lines.

Micro Levelling

Micro levelling was used to remove residual differences with a long wavelength along line and short wavelength across line. Application of the micro levelling process removed the streaks that were sometimes visible when using various grid enhancements.

Digital Terrain Data

After verification the radar altimeter height was subtracted from the GPS height and the Geoid – Ellipsoid separation correction applied to give the elevation of the terrain above mean sea level.

Gridding and Inspection

The digital terrain data was gridded and grid image enhancements were computed and displayed on screen. These were viewed also with the aid of crossline sun angles and inspected for inconsistencies and errors and appropriate corrections were made if required.

4.2.2 Magnetic Data

The raw un-edited magnetic data was checked to identify noise and spikes. Single reading spikes were manually edited.

Magnetic Diurnal Data

The diurnal data was then checked and corrected for spikes. Single reading spikes were manually edited and multiple erroneous readings flagged as invalid.

Diurnal Correction

The synchronized digital diurnal data collected by the base station was first subtracted from the corresponding airborne magnetic readings to calculate a difference. The resultant difference was then subtracted from the base value to produce diurnally corrected magnetic data.

Parallax Correction

A parallax error correction as described in section 3.3 was applied to the coordinate data.

IGRF correction

The magnetics data has been corrected for the regional gradient by subtracting the calculated IGRF (2005 model) computed continuously over the whole area. The calculation of these corrections used the GPS flying height.

Tie Line Levelling

A crossover program was used to compute the magnetic difference between each tie line and the traverse line intersection. These differences were then applied to level the traverse lines to the tie lines.

Micro Levelling

Micro levelling was used to remove residual differences with a long wavelength along line and short wavelength across line. Application of the micro levelling process removed the streaks that were sometimes visible when using various grid enhancements.

Gridding and Inspection

The magnetic data was gridded and grid image enhancements were computed and displayed on screen. These were also viewed with the aid of crossline sun angles and inspected for inconsistencies and errors and appropriate corrections were made if required.

4.2.3 Radiometric Data.

IAEA Processing.

The processing of the radiometric data is summarized below.

- Apply the deadtime correction.
- Energy recalibrate the 256 channel spectra and re-window the data.
- Noise adjusted Singular Value Decomposition (NASVD).
- Remove spikes from the altimeter, temperature and pressure values.
- Correct radiometric data to standard temperature and pressure.
- Remove the aircraft background, apply the cosmic correction, remove radon, apply the stripping values and finally apply the height correction.

Deadtime correction.

The GR-820 spectrometer requires a finite time to process each pulse from the detectors. The deadtime of the GR-820 is less than 5 microseconds per detector and this correction was applied.

Energy Recalibration.

Spectra analysis was performed on each line of data and the position of the thorium and potassium peak positions determined and compared to their theoretical positions. The original spectra data was then mapped to the correct peak positions and new windowed data created for each of the standard IAEA windows as follows.

Window	Peak Energy (KeV)	Energy Window (KeV)								
Total Count		410	-	2810						
Potassium	1460	1370	-	1570						
Uranium	1760	1660	-	1860						
Thorium	2615	2410	-	2810						
Cosmic		3000								

256 Channel Noise Reduction.

The 2 most common processing methods are: -

- Noise adjusted Singular Value Decomposition (NASVD). This was developed specifically for radiometric processing.
- Maximum Noise Fraction (MNF). This was developed for removing noise from satellite images and subsequently used in radiometric processing.

Both methods use Principal Component Analysis (PCA) with the only difference being in the estimation of noise in the raw spectra and subsequent scaling before PCA.

We have implemented and extensively used both methods but prefer NASVD because it is simpler, requires one less pass of the data and less observations for a good join when adjacent data sets are merged. However the 2 methods give almost the same result and both work well.

Careful analysis of the eigenvalues and eigenvectors of the PCA is required to ensure the process has worked correctly. We use the 7 most significant principal components to reduce the data with the remainder considered to be noise. If this is not the case, as seen from eigenvalue and eigenvector plots, then there is a problem with the data. So this is an excellent quality control tool as well as a noise reduction method. There are strong theoretical reasons for this approach and if less than 7 components are used some signal is likely to be removed. On large surveys we have found it is best to use 7 components globally rather than having to make difficult decisions for different segments of the survey as this provides a globally consistent result.

As final proof the method has worked correctly, residual line profiles and images of potassium, uranium and thorium must confirm that no signal is present. Also the ternary potassium, uranium and thorium image must be sharp. If signal has been removed this image will be blurred.

Standard Temperature and Pressure correction.

The data was converted to effective height at standard temperature and pressure using the expression:

STP(alt) = Radar Altitude * (Pressure/1013) * (273 / (Temperature+273))

Cosmic Correction

The aircraft background radiation was removed by subtracting the aircraft background values determined in the Yarra Yarra test from the Total Count, Potassium, Uranium and Thorium windows. The effect of cosmic radiation was removed from each window by multiplying the cosmic channel by the cosmic stripping factor for each window and subtracting the result from the window data.

Stripping

The radiometric spectra of potassium (K), uranium (U) and thorium (Th) series overlap. To evaluate of any one spectral window, which is designed to detect one radioelement, requires removal of the spectral overlap. This process of removal of the spectral overlap is known as stripping. The stripping procedure uses spectral stripping ratios determined experimentally using concrete calibration pads of known K, U and Th concentration.

Parallax Correction

A parallax error correction as described in section 3.3 was applied to the coordinate data.

Tie Line Levelling

A crossover program was used to compute the radiometric difference between each tie line and the traverse line intersection. These differences were then applied to level the traverse lines to the tie lines.

Micro Levelling

Micro levelling was used to remove residual differences with a long wavelength along line and short wavelength across line. Application of the micro levelling process removed the streaks that were sometimes visible when using various grid enhancements.

Gridding and Inspection

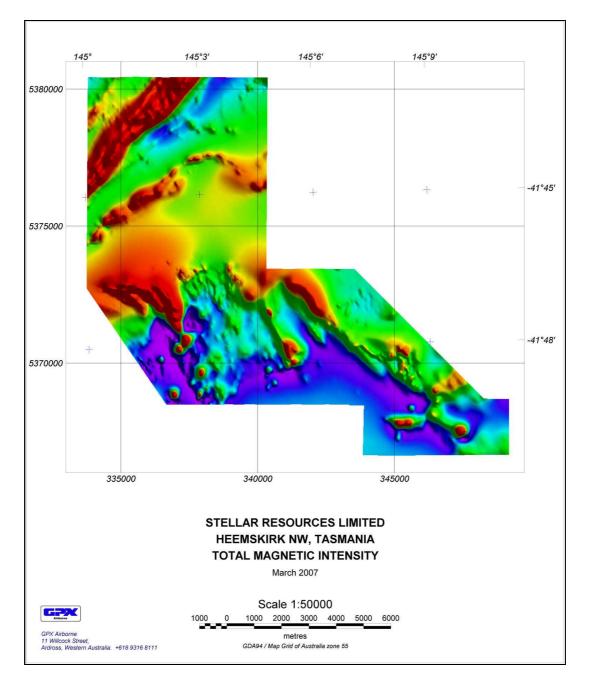
The radiometric data was gridded and grid image enhancements were computed and displayed on screen. These were also viewed with the aid of crossline sun angles and inspected for inconsistencies and errors and appropriate corrections were made if required.

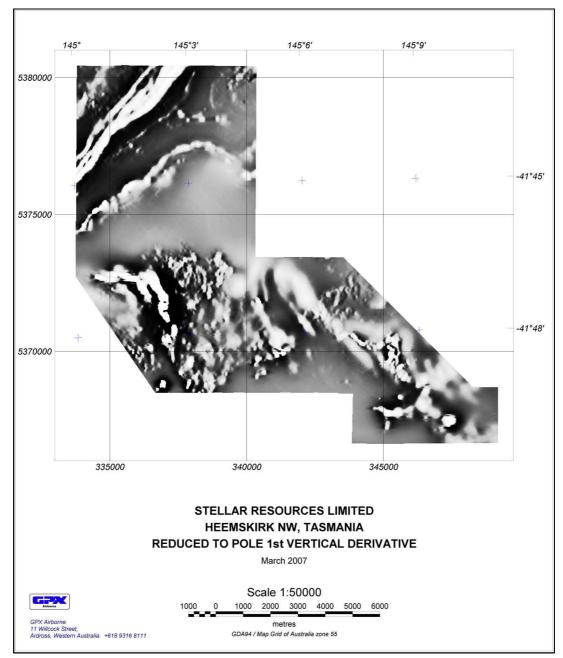
4.2.4 Digital Archives

The final digital data was written out as a flat ASCII located data file. The format and channel description can be found in Appendix A. Grids of the final data were created in ERMapper format.

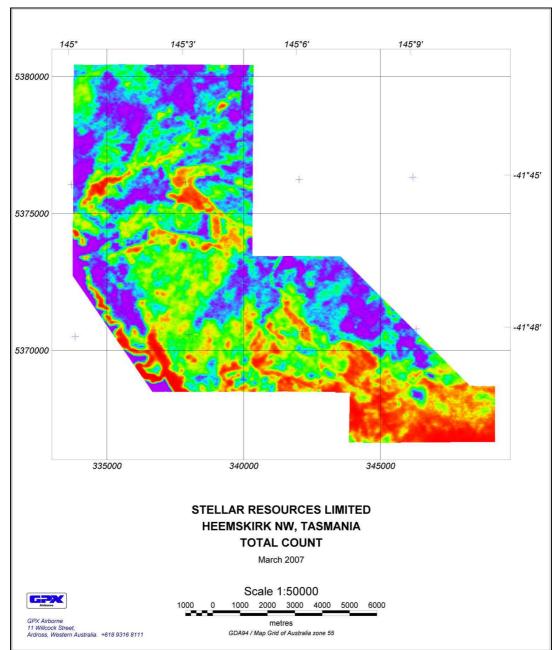
5 IMAGES







5.2 REDUCED TO POLE 1ST VERTICAL DERIVATIVE IMAGE



5.3 TOTAL COUNT IMAGE

6 CONTRACTOR INFORMATION



GPX Airborne

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Telephone: +618 9316 8111 Fax: +618 9316 8033

Web: http://www.gpxair.com.au/

7 APPENDIX A: FINAL LOCATED DATA FORMAT.

7.1 MAGNETIC DATA

GENERAL

Project Survey area Located data type	2273 Heemskirk, NW 0.1 Second Fin				
Surveyed by Job number Processed by Creation date	GPX AIRBORNE F 2273 GPX AIRBORNE F March 2007				
SURVEY SPECIFICATIONS					
Survey flown Traverse line spacing Traverse line direction Tie line spacing Tie line direction	February 2007 50 metres 090-270 degree 500 metres 000-180 degree				
Survey height	80 metres				
LOCATED DATA FORMAT					
Variable	Units	Undefined	From	То	Format
Line number Easting (Transverse Mercator) Northing (Transverse Mercator) Fiducial Flight number Direction (1=E, 2=N, 3=W, 4=S) Date (YYYYMMDD) Time (GPS) Longitude (WGS84) Latitude (WGS84) Radar altimeter GPS altitude Raw magnetics Post compensated magnetics Diurnal Final magnetics Pressure Temperature Raw total count Raw potassium Raw uranium	<pre>metres metres s) seconds degrees degrees metres metres nT nT nT nT nT millibars degrees C cps cps cps</pre>	9999999 9999999 999999 99 9999 99 9999 999 9999	1 9 20 31 40 44 46 55 64 75 86 93 100 110 120 130 140 147 152 159 164	8 19 30 39 43 45 54 63 74 85 92 99 109 129 129 139 146 151 158 163 168	$\begin{matrix} 18\\ F11.2\\ F11.2\\ F9.2\\ I4\\ I2\\ I9\\ F9.2\\ F11.6\\ F11.6\\ F11.6\\ F7.1\\ F7.1\\ F10.3\\ $
Raw thorium Raw cosmic Final total count Final potassium	cps cps cps	9999 99999.9 99999.99 99999.99	169 174 179 188	173 178 187 196	F5.0 F5.0 F9.2 F9.2
Final uranium Final thorium Final DTM	cps cps metres	99999.99 99999.99 99999.9	197 206 215	205 214 222	F9.2 F9.2 F8.1

DATA PROCESSING

COORDINATE DATA

All lines are scissored to the following rules:

- 1) A 'smooth' edge outside the area boundary.
- 2) Maximum line overlap of 0 fiducials within the area boundary.

The local projection is a UTM projection based on the

WGS84 spheroid with a central meridian of 147 East degrees. System parallax of 0.6 fiducial has been removed.

MAGNETIC DATA The magnetic data has been corrected for regional gradient by subtraction of IGRF model 2005 computed continuously over the whole area based on the GPS height. Diurnal magnetic variations have been removed. System parallax of 1.6 fiducial has been removed. Tie-line levelling has been applied. Microlevelling has been applied. A base value of 61900 nT has been added to the data.

RADIOMETRIC DATA

Raw channel data provided has been energy calibrated NASVD has been applied to channel data prior to windowing System parallax of 0.0 fiducial has been removed. Height attenuated to 80m AGL Airborne radon has been removed

AIRCRAFT BACKGROUND		UNITS	
Total Count	81.67	cps	
Potassium	23.35	cps	
Uranium	0.59	cps	
Thorium	0.67	cps	
COSMIC STRIPPING RATIOS		_	
Total Count	0.811201		
Potassium	0.044663		
Uranium	0.038646		
Thorium	0.043791		
COMPTON STRIPPING RATIOS			
alpha	0.250		
beta	0.400		
gamma	0.810		
a	0.060		
HEIGHT ATTENUATION COEFFICIENT			
Total Count	0.006928	L	
Potassium	0.009043	-	
Uranium	0.007186	-	
Thorium	0.006892	per metre	
WINDOW ENERGY LEVELS		High Energy	_
Total Count	410.0	2810.0	keV
Potassium	1370.0		keV
Uranium	1660.0		keV
Thorium	2410.0	2810.0	keV

DIGITAL TERRAIN MODEL DATA

DIGITAL TERRAIN MODEL CALCULATION

The radar altimeter data was subtracted from the GPS heights to provide a digital elevation model which is height above the WGS84 spheroid. Using interpolation on the 120 second DMA Geoid model, a correction was computed and subtracted from the WGS84 data to convert to height above the geoid.

DATA RELIABILITY

This Digital Terrain Model (DTM) has been computed from data generated during the course of an airborne geophysical survey flown at a nominal spacing of 50m and data has been interpolated between such lines. Every effort has been made to make this model a useful general reference. No guarantee can be made that this model is a true representation of height above sea level as it can contain radar altimeter responses from buildings and in some instances dense timber. Users of the product should be aware of the topographic limitations mapped herewithin. Do not use this DTM for navigation purposes.

7.2 RADIOMETRIC DATA

GENERAL

Project Survey area Located data type	2273 Heemskirk, NW 1 Second Radic				
Surveyed by Job number Processed by Creation date	GPX AIRBORNE F 2273 GPX AIRBORNE F March 2007				
SURVEY SPECIFICATIONS					
Survey flown Traverse line spacing Traverse line direction Tie line spacing Tie line direction	February 2007 50 metres 090-270 degree 500 metres 000-180 degree				
Survey height	80 metres				
LOCATED DATA FORMAT					
Variable	Units	Undefined	From	То	Forma
Line number Easting (Transverse Mercator) Northing (Transverse Mercator Fiducial Flight number Direction (1=E, 2=N, 3=W, 4=S Date (YYYYMMDD) Time (GPS) Longitude (WGS84) Latitude (WGS84) Radar altimeter Pressure Temperature Raw total count Raw potassium Raw uranium Raw uranium Raw thorium Raw cosmic Final total count Final potassium Final uranium)metres) seconds degrees degrees metres millibars degrees C cps cps cps cps cps cps cps cps cps cps	9999999 9999999.99 999999.99 999 9999999	1 9 20 31 40 44 46 55 64 75 86 93 100 105 112 117 122 127 132 141 150	8 19 30 39 43 45 54 63 74 85 92 99 104 111 126 131 140 149 158 167	<pre>I8 F11.2 F11.2 F9.2 I4 I2 I9 F9.2 F11.6 F11.6 F7.1 F7.1 F7.0 F5.0 F5.0 F5.0 F5.0 F5.2 F9.2 F9.2 F9.2 F9.2 F9.2 F9.2</pre>
Final thorium Raw 256 channel data Energy calibrated 256 channel	cba cba	99999.99 999 9999.9	159 168 1192	167 1191 2983	F9.2 I4 F7.1

DATA PROCESSING

COORDINATE DATA

All lines are scissored to the following rules:

- 1) A 'smooth' edge outside the area boundary.
- 2) Maximum line overlap of 0 fiducials within the area boundary.

The local projection is a UTM projection based on the WGS84 spheroid with a central meridian of 147 East degrees. System parallax of 0.6 fiducial has been removed.

RADIOMETRIC DATA

NASVD has been applied to channel data prior to windowing System parallax of 0.0 fiducial has been removed.

Format

Height attenuated to 80m AGL Airborne radon has been removed

AIRCRAFT BACKGROUND		UNITS
Total Count	81.67	cps
Potassium	23.35	cps
Uranium	0.59	cps
Thorium	0.67	cps
COSMIC STRIPPING RATIOS		
Total Count	0.811201	
Potassium	0.044663	
Uranium	0.038646	
Thorium	0.043791	
COMPTON STRIPPING RATIOS		
alpha	0.250	
beta	0.400	
gamma	0.810	
a	0.060	
HEIGHT ATTENUATION COEFFICIENT		
Total Count	0.006928	per metre
Potassium	0.009043	per metre
Uranium	0.007186	per metre
Thorium	0.006892	per metre
WINDOW ENERGY LEVELS	Low Energy	High Energy
Total Count	410.0	2810.0
Potassium	1370.0	1570.0
Uranium	1660.0	1860.0
Thorium	2410.0	2810.0

keV keV keV keV

C	ient	GPXAIR	Job No.	Area(s)		Job Name)	F	lying Base		Airc	raft Type	(s)	Crew C	Contact P	hone No	Crew	Contact	Sat No
Stellar F	Resouces	22	273	1	Hee	mskirk - Ta	smania	Strah	an - Tasm	ania	Ce	essna 21	0	+61 (0)438 9	27 443	+88	162146	62769
Aircraft:		AOM	Bob Blizza	ard		FPM	Jeff Ibbottso	n	Field Oper	rators	Don Cople	әу		Pilots		Guy Nash	1	Vincent	Wong
			1		ĸ	ilometres						Aircraft	Timo				1		
	Date	Flt	Prod	Scrub	Reflight	Total	Flown to	Remain	Prod	Scrub	Turns	Ferry	Cals	Cals	Mob	Total	Hrs to	Sta	ndby
					Ũ	Planned	date					-	Daily	Setup			100 Hrly	Hrs	Reason
Sat	17/02/07		0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0		0.0			0.0	0.0	0.0	0.0	
Sun	18/02/07		0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0		0.0			0.0	0.0	0.0	0.0	
Mon	19/02/07		0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0		0.0			0.0	0.0	0.0	0.0	
Tue	20/02/07		0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0		0.0			0.0	0.0	0.0	0.0	
Wed	21/02/07		0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0		0.0			0.0	0.0	0.0	0.0	
Thu	22/02/07		0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0		0.0			0.0	0.0	0.0	0.0	
Fri	23/02/07	01	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0		0.0			8.2	8.2	80.0	0.0	
	Totals:		0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0		0.0			8.2	8.2		0.0	
											-								
		Chargea	ble Lost Ti	me (Hrs):			Non-Charg	geable Lost	Time (Hrs):		J								
	Date	Julian	GPXAIR	Other	Accom'	Vehicles	Vehicles Aircraft Maint' Equip Ma					General / Processing / QC Comments							
		Day	Crew	Crew	(Rooms)		Hrs	S/U	Hrs	S/U						g			
Sat	17/02/07	48																	
Sun	18/02/07	49																	
Mon	19/02/07	50																	
Tue	20/02/07	51																	
Wed	21/02/07	52																	
Thu	00/00/5=										I								
Thu	22/02/07	53																	
Fri	23/02/07	54	1	3	4						Crew door	arted Ten	ant Creck	for Strop	an Tasm	ania. VH-		nighted	'n
	20/02/07	- 54		5	-							Victoria.					WINN OVER	ngnteu	
	Totals:						0.0		0.0										
											-		(010)	- "					
	KEY:	mbore/da	ta into <u>blu</u>	o markod	soctions o	nly				Moh =	<pre>= Safet (Mob/Dem</pre>					(TBM) ch			
						should not b	o obongod				(Mob/Dem t/Equipmer						•		
							e changed.										hor doco	rintion	
	relly = 0	perations	s base to s	survey are	a & return	umes.				Gener	al/Processi	ing/QC C	omment	s snouid i		basic weat	ner uesci	iption.	

8 APPENDIX B: WEEKLY PRODUCTION SUMMARY

			Job No.	Area(s)		Job Name	e	F	lying Base		Airc	raft Type	(s)	Crew C	ontact I	Phone No	Crew	Contac	t Sat No		
Stellar F	Resouces	22	273	1	Heer	mskirk - Tasmania Strahan - Tasmania						Cessna 210 +61 (0)438 927 443					+881621462769				
Aircraft:		AOM	Bob Blizza	ard		FPM	Jeff Ibbottso	n Field Operators		Don Copley			Pilots Guy Nas		Guy Nas	n	Vincent Wong				
					ĸ	ilometres						Aircraft	Timo				1				
1	Date	Flt	Prod	Scrub	Reflight	Total	Flown to	Remain	Prod	Scrub	Turns	Ferry	Cals	Cals	Mob	Total	Hrs to	St	andby		
	Duito			00.00		Planned	date			00.45		,	Daily	Setup		, orall	100 Hrly	Hrs	Reason		
Sat	24/02/07	02	0.0	0.0	0.0	2,438.2	0.0	2,438.2	0.0	0.0		0.0			2.9	2.9	77.1	0.0			
Sun	25/02/07		0.0	0.0	0.0	2,438.2	0.0	2,438.2	0.0	0.0		0.0			0.0	0.0	77.1	0.0			
Mon	26/02/07	03/04	72.0	0.0	0.0	2,438.2	72.0	2,366.2	1.1	0.0		2.2		1.0	0.0	4.3	72.9	3.7	Fuel		
Tue	27/02/07	04 / 05	932.0	932.0	0.0	2,438.2	72.0	2,366.2	7.0	7.0		1.4			0.0	8.4	64.5	0.0			
Wed	28/02/07	06	479.2	0.0	0.0	2,438.2	551.2	1,887.0	3.7	0.0		0.2			0.0	3.9	60.6	4.1	Weather		
Thu	01/03/07	07	568.6	0.0	0.0	2,438.2	1,119.8	1,318.4	3.9	0.0		0.2			0.0	4.1	56.7	3.9	Weather		
Fri	02/03/07	08 / 09	726.6	223.7	330.0	2,438.2	1,952.7	485.5	6.5	3.4		0.4			0.0	6.9	49.8	0.0			
	Totals:		2,778.4	1,155.7	330.0	2,438.2	1,952.7	485.5	22.2	10.4		4.4			2.9	30.5		11.7			
											-										
		Chargeat	ole Lost Ti	me (Hrs):		3.7	Non-Charg	geable Lost	Time (Hrs):	8.0	J										
1	Date	Julian	GPXAIR	Other	Accom'	Vehicles	Aircra	Aircraft Maint' Equip Maint'				General / Processing / QC Comments									
		Day	Crew	Crew	(Rooms)		Hrs	S/U	Hrs	S/U						-					
Sat	24/02/07	55	1	3	4	1					Aircraft VH	H-MNN arı	rived Stra	han, Tasr	nania in	late pm. J	. Ibbottsor	and D.	Copley		
											arrived fro	m Burnie	in late pr	n. No fuel	availabl	e at this tin	ne (weeke	nd).			
Sun	25/02/07	56	1	3	4	1					No flights	undertake	n, Base I	Mag was s	set up an	id still no fu	iel.		TBM		
Mon	26/02/07	57	1	3	4	1					2 Elights	02 was Co	omp-box :	and ferry t	o Launo	eston, to re	efuel 0.3 v	vas reco			
	20/02/01	07		Ŭ	-						U .		1 A A A A A A A A A A A A A A A A A A A			fuel in Bu					
Tue	27/02/07	58	1	3	1	1			7.0	U	2 Flights -		-					2060 to	1370.		
				-								bed due									
Wed	28/02/07	59	1	3	4	1					-					d and rain	(low visibil	ty) in pn	n.		
Thu	01/03/07	60	1	3	4	1					1 flight - 0	7, 690 to ⁻	1200. Hig	gh winds a	and low l	evel cloud	pm.				
Fri	02/03/07	61	1	3	4	1			3.4	U	2 flights -	08. 1210 t	o 1370 a	nd 10010	to 10310) and flt - 0	9, 1380 to	1870.			
				_						_	Partial refl						.,				
	Totals:						0.0		10.4												
	All areas	s marked i	n black ar	e self calo	sections o culating & s a & return	should not b	e changed.			Aircrat	= Safet (Mob/Dem t/Equipmer al/Processi	ob) initial nt Maint':	flying tir S = sche	ne to proj eduled / U	ject & fro = unsc	heduled.	t.				

Client Stellar Resouces		GPXAIR	Job No.	Area(s)	Job Name Heemskirk - Tasmania			Flying Base Strahan - Tasmania			Aircraft Type (s) Cessna 210			Crew C	ontact F	hone No	Crew	Crew Contact Sat No		
		22	273	1										+61 (0)438 927 443			+881621462769			
Aircraft:		AOM	Bob Blizza	ard		FPM	Jeff Ibbottsor	n	Field Ope	rators	Don Copl	еу		Pilots		Guy Nas	'n	Vincent	Wong	
·						Kilometres				Aircraft Time							1		-	
1	Date	Flt	Flt Prod	Scrub	n Reflight	Total	Flown to	Remain	Prod	Scrub	Turns	Ferry		Cals	Mob	Total	Hrs to	St	andby	
	Dute		mou	00100	Reingin	Planned	date	Remain	110u	corus	runio	1 cmy	Daily	Setup	mob	Total	100 Hrly	Hrs	Reason	
Sat	03/03/07	10	52.8	0.0	0.0	2,438.2	2,005.5	432.7	0.8	0.0		0.2		_	0.0	1.0	48.8	7.0	Process	
Sun	04/03/07	11/12	432.7	0.0	559.2	2,438.2	2,438.2	0.0	5.1	0.0		0.7			0.0	5.8	43.0	2.2	Process	
Mon	05/03/07	13/14	0.0	0.0	266.5	2,438.2	2,438.2	0.0	3.6	0.0		0.4			0.0	4.0	39.0	0.0		
Tue	06/03/07	0	0.0	0.0	0.0	2,438.2	2,438.2	0.0	0.0	0.0		0.0			0.0	0.0	0.0	0.0		
Wed	07/03/07	0	0.0	0.0	0.0	2,438.2	2,438.2	0.0	0.0	0.0		0.0			0.0	0.0	0.0	0.0		
Thu	08/03/07	0	0.0	0.0	0.0	2,438.2	2,438.2	0.0	0.0	0.0		0.0			0.0	0.0	0.0	0.0		
Fri	09/03/07	0	0.0	0.0	0.0	2,438.2	2,438.2	0.0	0.0	0.0		0.0			0.0	0.0	0.0	0.0		
	Totals:		485.5	0.0	825.7	2,438.2	2,438.2	0.0	9.5	0.0		1.3			0.0	10.8		9.2		
	i	Chargoak	ole Lost Ti	mo (Hrs).		9.2	Non-Chard	eable Lost	Timo (Hre):	T	٦									
		Chargean		ille (filis).		5.2	Non-Charg		rine (rits).	I	1									
1	Date	Julian	GPXAIR	Other	Accom'	Vehicles	Aircraft Maint'		Equip Maint'					General	/ Proces	sing / QC	Commen	its		
		Day	Crew	Crew	(Rooms)		Hrs	S/U	Hrs	S/U						•				
Sat	03/03/07	62	1	3	4	1					1x Production Flight - Flt 10, Lines 1880 to 1950. Waiting on processing results.									
Sun	04/03/07	63	1	3	4	1					2 x Production Flights - Flt 11 / Flt 12. Waiting on processing results.									
Mon	05/03/07	64	1	3	1	1					2x Production Flights - Flt 13 / Flt 14. VH-MNN departs Strahan, with J.I and G.N									
Tue	06/03/07	65									Ground Vehical Departs Strahan, with D.C and V.W.									
Wed	07/03/07	66																		
Thu	08/03/07	67																		
Fri	09/03/07	68																		
	Totals:						0.0		0.0											
	KEY: Enter numbers/data into <u>blue marked</u> sections only. All areas marked in black are self calculating & should not be changed. Ferry = operations base to survey area & return times.									Aircrat	 Safety Meeting (SM) / Toolbox Meeting (TBM) check box. Mob = (Mob/Demob) initial flying time to project & from project. Aircratt/Equipment Maint': S = scheduled / U = unscheduled. General/Processing/QC Comments should include basic weather description. 									